

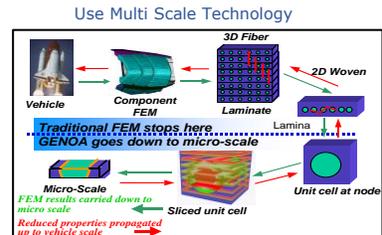
## GENOA DISCRETE COHESIVE ZONE MODELING (DCZM)

- DCZM is part of a High-Fidelity Solution Flow Process where the goal is to assess Damage and Fracture Evolution.
- Requires test calibration to determine the best combination of cohesive strength and cohesive stiffness.
- Fracture mechanics based approach for the progressive crack growth analysis integrated into GENOA-PFA
- Based on the linear spring elements
- Insensitive to FEM mesh size. Avoids the use of "singular" crack elements. As a result, extensive mesh preparation is eliminated
- Is computationally efficient due to the use of the node-based displacements and forces, which does not affect the problem size
- Requires a fracture path to be predetermined based on either: 1) experimental testing; 2) a preliminary GENOA/PFA analysis; or 3) the user experience. The predetermined path can be effectively prescribed using the GENOA GUI
- Can be used for computing strain energy release rate in linear elastic materials
- Requires fracture toughness data as input. The fracture toughness test data can be obtained from testing, material handbooks, or from GENOA-FTD.
- Obtain solutions with NASTRAN, ABAQUS, ANSYS, OPTISTRUCT, MHOST, xDYNA, LS DYNA, RADIOSS
- Augments finite element analysis (FEA) with multi-scale composite mechanics.

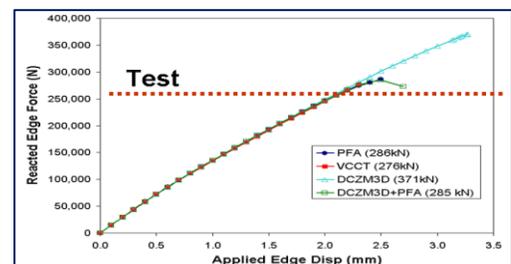
GENOA DISCRETE COHESIVE ZONE MODELING allows engineers to perform a fracture mechanics based approach to characterize the durability of composite structures. This analysis determines crack propagation mechanisms in composites, and predicts facesheet-core delamination in sandwich materials. The simulation can be coupled with GENOA PFA to determine laminate and ply damage (types: fiber, matrix, delamination – transverse shear, interlaminar shear, relative rotation, fiber microbuckling, fiber pullout), damage and fracture initiation, energy absorbed, and residual strength. GENOA PFA will accurately predict the behavior of advanced composite laminates (2-D/3-D) considering effects of (1) defects, voids, fiber waviness, (2) micro-crack density (leakage, stiffness reduction), (3) residual stresses (winding, curing).

## GENOA VIRTUAL CRACK CLOSURE TECHNIQUE

- ✓ **Supports full breadth of 2D/3D composite architectures**
  - Laminated Tape Lay-Up, Polymer, Metals, Ceramics
  - Fiber Architecture (Woven, Triaxial, Harness Satin Weave, Braided, and Stitched)
  - Fiber Coating (InterPhase), Effects of manufacturing defects and residual stresses
- ✓ **Determines crack propagation mechanisms in composites**
  - Apply to 2D line crack and 3D surface cracks
- ✓ **Couple with GENOA-PFA to**
  - Determine composite damage
  - Improve accuracy
  - Support All GENOA\_PFA Failure Criteria (In-built and User Defined)
  - Support Detailed Micromechanical Degradation
  - Support All Types of Service Loading
  - Laminate and Ply Damage initiation and propagation to final failure
  - Damage types (fiber, matrix, several delamination types)
  - Offer load displacement shedding (after peak load)
  - Change ply layups to meet design requirements
  - Residual strength behavior (TAI, CAI, FAI)
  - Export Damage/Residual Stresses use in another simulation/solver
- ✓ **Supports Service Loading**
  - Quasi Static Fatigue and Random Fatigue (low, high, random, PSD, two stage)
  - **Export Damage/Residual Stresses use in another simulation/solver**
  - Change boundary conditions/solver/ analysis type
  - Static or Impact to static/fatigue/creep (any combination and sequence)
- ✓ **Includes Tutorials/Solutions**



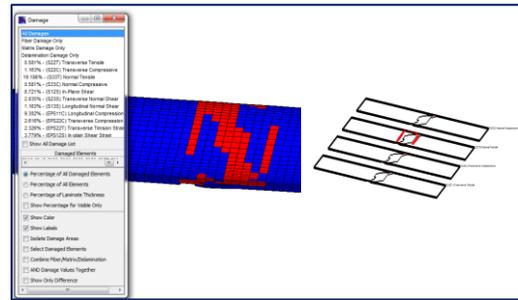
Failure prediction vs test for 3 stringer stiffened panel showing composite damage and disband.



## Key Benefits

- Efficiently solves crack propagation problems for aerospace, automotive and other structural applications.
- Predicts crack propagation mechanisms in composites.
- Predicts facesheet-core delamination in sandwich materials.
- Provides accurate failure analysis of adhesively bonded joints.
- Crack growth uncertainty in direction predicted
- Detects critical crack propagation and arrest in pipelines.
- Can be applied to interface failure analysis in MEMS (Micro Electro Mechanical Systems).
- Reduce physical test by over 65-70% thus saving significant cost  
Ease of use, results verified with test data for class of materials:  
Polymer: chopped, continuous, thermoset, thermoplastic, elastomer.  
Ceramic Metals: Fracture Toughness, fatigue crack growth. Nano Hybrid Composite (Glare)

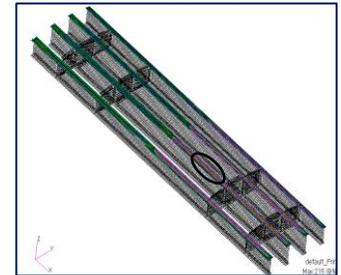
Accumulated damage and fracture at the crack tip for 3D DCZM



Bond line failure can easily be modeled with GENOAs DCZM elements. Wing box from Global Hawk.

## User Friendliness

- Graphic User Interface (GUI) is easy to learn with navigation tutorials and videos. Manages multiple projects, input and output for material characterization
- Quick import/export of material properties and laminate layups with commonly used third-party FE Solvers and UMATs: NASTRAN (.bdf), ABAQUS (.inp), ANSYS (.cdb), RADIOSS (.rad), LSDYNA (.k) and Optistruct (.fem)
- **Easy creation and editing of composite laminates. Quickly study multiple designs.**



## System Requirements

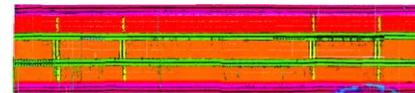
- Windows XP/Vista/7/8/10 (64-bit) or Linux (64-bit)
- Java 1.7 minimum Runtime Libraries
- Java3D 1.5

## Minimum Configuration

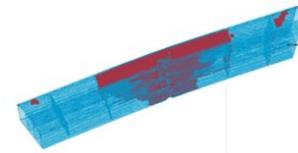
With the minimum configuration, performance and functionality may be less than expected.

- 1 GHz or higher CPU, 4GB RAM, 10GB disk space

Account for Initial Debond/Defects



Combining GENOA PFA and DCZM results in Bond and Laminate Damage Prediction (Mixed Mode)



## Step By Step Instructions

**Discrete Cohesive Zone Model (DCZM3D)**

Case Description: Crack growth simulation by fracture mechanics approach for nonlinear materials

Example Location: **Tutorials > Fracture > DCZM3D**

Model Description:



Beam length: 100mm, Beam width: 20mm and Beam thickness: 1.3mm  
Laminated Composites similar to Carbon/PEEK (AS4/APC2)

Material Description

Objective of Analysis: Determine the ultimate (failure) load and associated crack growth.

ASTM Number: N/A

Control Type: Displacement Control

Analysis Type: Static

Solution: \*STATIC 3 (Analyze Using Updated Geometry = False [Total Lagrangian method])

Input Requirements: Assign Fracture toughness, cohesive strength, and cohesive stiffness

FEA Solver: **ABAQUS**  
**NASTRAN**

Output from Analysis: End load vs. displacement curve; Crack growth animations

Summary of Results: **Load vs. displacement curve:**  
Nonlinear up to the maximum load and nonlinear portion of the load drop.

**Introduction**

DCZM (Discrete Cohesive Zone Model) is developed in GENOA/PFA to study progressive crack growth. T<sub>11C</sub> and T<sub>112C</sub>. The crack path should be pre-described has duplicated nodes. GENOA GUI will assist. The current DCZM3D works with NASTRAN and ABAQUS FE solvers.

## Tutorials with Solutions and Code Verifications

**Code Verifications**

- ABAQUS
- MHOST
- NASTRAN
  - DCZM
    - DCZM Center Crack Example 1
    - DCZM Center Crack Example 2
    - DCZM Compact Tension
    - DCZM Double Cantilever Beam (1200 Elements)
    - DCZM Double Cantilever Beam (400 Elements)
    - DCZM Inclined Crack

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More of Alpha STAR's Test Validated products:

MCQ: Composites, Ceramics, Metals, Nano, Chopped  
GENOA: PFA, PFDA, UAB, URD, ABS, PCP, PA, Quasi Static Fatigue & Random Fatigue, Harmonic & PSD Fatigue, Fatigue with Fracture Mechanics, PFA\_AGING, VCCT, DCZM, Filament Winding, Jobspooler, GENOA\_CLOUD