

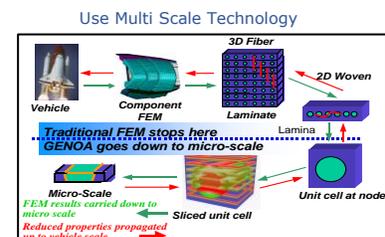
## GENOA HARMONIC FATIGUE AND POWER SPECTRUM DENSITY

- Test Validated™ Solutions
- Obtain fatigue solutions with NASTRAN, use results in all other solvers: ABAQUS, ANSYS, OPTISTRUCT, MHOST, xDYNA, LS DYNA, RADIOSS
- Augments finite element analysis (FEA) with multi-scale composite mechanics.
- Damage tracking and fracture to determine all stages of damage evolution under harmonic and power spectral density fatigue.
- Predict and simulate all 5 stages of the damage process.
- Switch solvers/boundary condition/analysis type before or after simulation and keep residual damage and stresses.
- Calculates crack density, micro-cracks in the matrix, delamination within the plies, and fiber failure in tension and compression including micro-buckling.
- Account for defects (void shapes and sizes), fiber waviness, and residual stresses – **Voids/Defects Will Reduce Fatigue Life.**
- The damage tracking is done by identifying and accumulating damage at the "root cause" of the composite in matrix and fiber using dedicated physics based damage and failure criteria.

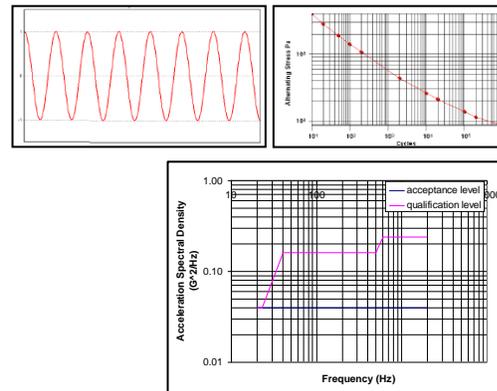
GENOA HARMONIC FATIGUE AND PSD FATIGUE allows engineers to perform a GENOA Multi Scale Progressive Failure Analysis and characterize the fatigue behavior of composite structures under harmonic, spectrum, or PSD loads. This analysis determines: laminate and ply damage (types: fiber, matrix, delamination – transverse shear, interlaminar shear, relative rotation, fiber microbuckling, fiber pullout), damage and fracture initiation, energy absorbed, and residual strength. GENOA PFA will accurately predict the behavior of advanced composite laminates (2-D/3-D) considering effects of (1) defects, voids, fiber waviness, (2) micro-crack density (leakage, stiffness reduction), (3) residual stresses (winding, curing).

## GENOA HARMONIC FATIGUE AND POWER SPECTRUM DENSITY FATIGUE

- ✓ **Supports full breadth of 2D/3D composite architectures**
  - Laminated Tape Lay-Up, Polymer, Metals, Ceramics
  - Fiber Architecture (Woven, Triaxial, Harness Satin Weave, Braided, and Stitched)
  - Fiber Coating (InterPhase), Effects of manufacturing defects and residual stresses
- ✓ **Determines composite damage**
  - Laminate and Ply Damage initiation and propagation to final failure
  - Damage types (fiber, matrix, several delamination types)
  - Change ply layups to meet design requirements
  - Residual strength behavior (TAI, CAI, FAI)
- ✓ **Supports Failure Criteria (In-built and User Defined)**
  - Translaminar (Matrix, Fiber, Ply)
  - Interlaminar/Delamination (Tension, Shear, Relative Rotation)
  - Interactive Strength (Tsai-Wu, Tsai-Hill, Puck, MDE, Hoffman, Hashin)
  - Interactive Strain- Strain Invariant Failure Theory (SIFT)
  - Maximum Stress, Maximum Strain, User Defined
- ✓ **Supports Service Loading**
  - Harmonic Fatigue and PSD Fatigue
  - **Export Damage/Residual Stresses used in another simulation/solver**
  - Change boundary conditions/solver/ analysis type
  - Static or Impact to static/fatigue/creep (any combination and sequence)
- ✓ **Includes Tutorials/Solutions**



Predict Damage From Harmonic and PSD Loading



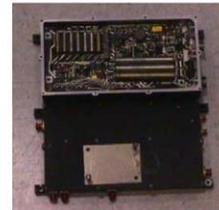
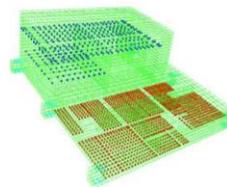
## Key Benefits

### HARMONIC:

- Evaluates the fatigue life of composite and metal structures subject to dynamic loads of constant amplitude.
- Determines the fatigue failure initiation site in composite and metal structures under the applied cyclic loading.
- For each stress level, the degree of cumulative damage incurred is calculated from the S-N curve, which can be derived from tests or found in the open literature.
- An efficient multi-factor interaction model (MFIM) can be used to account for a broad range of factors that affect the fatigue life of a structure such as temperature and surface finish.
- Can be used with GENOA-PA to account for the probabilistic nature of fatigue.

- The degradation S-N curve for a considered material.
- **POWER SPECTRUM DENSITY:**
- Evaluates various mode shapes for the fatigue life of composite and metal structures subject to dynamic loads of constant amplitude and vibration frequencies above 1/3 natural frequencies.
- Determines the fatigue failure initiation site in composite and metal structures under the applied cyclic loading.
- For each stress level, the degree of cumulative damage incurred is calculated from the S-N curve, which can be derived from tests or found in the open literature.
- An efficient multi-factor interaction model (MFIM) can be used to account for a broad range of factors that affect the fatigue life of a structure such as temperature and surface finish.
- Evaluate material S-N curve degradation with increase in modal frequencies.
- Determines degradation of natural frequencies.
- **Accurately evaluates the service life of composite or metal structures subject to static and dynamic loads of constant amplitude or PSD loading.**
- **Predicts the fatigue failure initiation site and its development in aerospace, automotive and other composite structural components.**
- **Accounts for the probabilistic nature of fatigue.**

GPS L3 Transmitter



**User Friendliness**

- Graphic User Interface (GUI) is easy to learn with navigation tutorials and videos. Manages multiple projects, input and output for material characterization
- Quick import/export of material properties and laminate layups with commonly used third-party FE Solvers and UMATS: NASTRAN (.bdf), ABAQUS (.inp), ANSYS (.cdb), RADIOSS (.rad), LSDYNA (.k) and Optistruct (.fem)
- **Easy creation and editing of composite laminates. Quickly study multiple designs.**

**System Requirements**

- Windows XP/Vista/7/8 or Linux (64-bit)
- Java 1.7 minimum Runtime Libraries
- Java3D 1.5

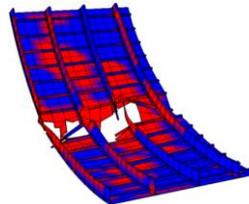
**Minimum Configuration**

With the minimum configuration, performance and functionality may be less than expected.

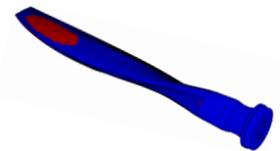
- 1 GHz or higher CPU, 4GB RAM, 10GB disk space

Step By Step Instructions

Stiffened Panel



Wind Blade



**Power Spectrum Density (PSD)**

Case Description: Cantilever beam with harmonic loading applied to its free end.

Example Location: **Tutorials > Fatigue > Power Spectrum Density**



Model Description: Shell Elements: 861  
Length = 400 mm  
Width = 200 mm

Objective of Analysis: Determine the degradation in natural frequency.

ASTM Number: N/A

Control Type: Load Control

Analysis Type: Dynamic

**Harmonic Fatigue**

Case Description: Cantilever beam with harmonic loading applied to its free end.

Example Location: **Tutorials > Fatigue > Harmonic Fatigue**



Model Description: Nodes: 156; Elements: 125  
Plate length: 5 inch, Plate width: 1.0 inch and Ply thickness: 0.1 inch

Objective of Analysis: Determine the number of cycles to failure.

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More of Alpha STAR's Test Validated products:  
MCQ: Composites, Ceramics, Metals, Nano, Chopped  
GENOA: PFA, PFDA, UAB, URD, ABS, PCP, PA, Quasi  
Static Fatigue & Random Fatigue, Harmonic & PSD  
Fatigue, Fatigue with Fracture Mechanics,  
PFA\_AGING, VCCT, DCZM, Filament Winding,  
Jobspooler, GENOA\_CLOUD