



MCQ Ceramics is a multi-scale material modeling, characterization & qualification software that predicts ceramic material performance. This tool enables engineers, designers and material specialists to quickly and accurately predict mechanical properties of advanced composite laminates considering environmental degradation effects. A building block reverse engineering approach is used to calibrate the “effective” constituent properties and architecture definitions. Using the effective properties to validate against the minimum available, MCQ Ceramics is independent of finite element modeling as it utilizes a unit cell approach.

Highlights:

- ✓ Supports full breadth of 2D/3D composite architectures
- ✓ Determines laminate properties such as Young’s Modulus
- ✓ Determines at lamina/laminate limit loads, stresses, and strain
- ✓ Supports failure criteria (in-built)
- ✓ Supports anisotropic matrix
- ✓ Generates material properties used with commercial FE software and has been validated for several classes of Ceramic Matrix Composites

- Oxide/Oxide
- SiC/SiC-CVI
- SiC/SiC-MI
- SiC/S200(SiNC)
- SiC/CAS

Key Benefits:

- ✓ Rapid assessment of properties
- ✓ Reduction in testing with cost savings
- ✓ Validated material library
- ✓ Strength allowable for reliability
- ✓ Identification of damage initiation/propagation to failure
- ✓ Identification damage/failure modes
- ✓ Results verified with test data

Modules

<ul style="list-style-type: none"> • Fiber/Matrix/Ply Calibration 	Reverse engineer the in-situ Young’s modulus, Poisson’s Ratio, Strength of transversely and isotropic fibers and isotropic matrix from tested in-plane ply properties.
<ul style="list-style-type: none"> • Architecture Fiber/Matrix Calibration 	Reverse engineer the braid architecture from coupon test data. It can be used for reverse engineering the in-situ Young’s modulus, Poisson’s Ratio, Strength of transversely and isotropic fibers and isotropic/anisotropic matrix from tested in-plane laminate properties.
<ul style="list-style-type: none"> • Ply Mechanics 	Verify ply properties from fiber and matrix constituent properties and variation in fiber and void volume ratio. The analysis relies on micro-mechanics theory.
<ul style="list-style-type: none"> • Ply Characterization 	Graphically verify the variation in ply properties and dominant failure zones with variation in fiber and void volume fraction and loading direction orientation with respect to the fibers.
<ul style="list-style-type: none"> • Laminate Mechanics 	Predict laminate level material properties using fiber/matrix, ply properties as input along with braid cards for fabric, woven or 3D architecture. Analysis relies on progressive failure analysis, micro-mechanics and classical laminate theory.
<ul style="list-style-type: none"> • Material Non-Linearity 	Predicts the in-situ matrix stress strain curve from in-plane shear ASTM standard test data. The analysis can be used to reverse engineer any fiber or ply nonlinearity as well. The analysis relies on progressive failure analysis, micro-mechanics and classical laminate theory.
<ul style="list-style-type: none"> • Progressive Failure 	Performs the analysis for the input using material degradation models and iterative process based on user input to ultimately predict the strength, modulus, and laminate and layer-by-layer damage evolution process.
<ul style="list-style-type: none"> • Design Failure Envelope 	Predicts failure envelope for lamina or laminates based on the chosen failure criteria after the above calibration process.
<ul style="list-style-type: none"> • Parametric Carpet Plot 	Predicts graphical representation of strength and other material properties of laminates containing symmetric and balanced plies in three different orientations.
<ul style="list-style-type: none"> • A- & B-Basis Allowables Calculator 	Predicts A- and B-basis strength allowables based on simple test data.
<ul style="list-style-type: none"> • A- & B-Basis Allowables Simulator 	Predicts A- and B-basis strength allowables based on material and fabrication uncertainty in the composite laminate material. You can directly enter the scatter from the unidirectional ASTM standard tests as the variation in the constituents.
<ul style="list-style-type: none"> • Manufacturing Defects Mechanics 	Predicts effect of manufacturing defects; fiber waviness on ply level properties.
<ul style="list-style-type: none"> • Manufacturing Defects Characterization 	Graphically predicts effect of manufacturing defects on ply level properties as a function of variation in fiber waviness.
<ul style="list-style-type: none"> • Constituent Fatigue Life 	Reverse engineer in-situ stress versus cycles to failure curve for the matrix using in-plane shear, transverse tension, and longitudinal tension fatigue life curves obtained typically from ASTM standard tests or literature.
<ul style="list-style-type: none"> • Progressive Fatigue Life 	Predicts fatigue life curve for laminates from ply or constituent level fatigue life input.
<ul style="list-style-type: none"> • Recession 	Recession occurs when surface material vaporizes due to high temperature and chemical reactions with the environment and material is removed when gas velocities are sufficient to carry material away. This is accounted by this capability by providing users with thickness degradation of the material with respect to time.
<ul style="list-style-type: none"> • Oxidation 	Oxidation occurs when the composite is infiltrated by oxygen/moisture through diffusion. For example SiC reacts to form SiO ₂ , which eats away fibers. In oxidation initially weight increases due to diffusion/reaction, followed by weight loss due to volatilization. This is accounted by this capability by providing users with thickness degradation of the material with respect to time.
<ul style="list-style-type: none"> • Strain Rate Effect 	Calculates Strain Rate of materials